

Toughening for PNTAB Dr. B. Parkinson

## **Toughening** A part of **PTA** Goal: <u>Assured PNT</u> A discussion of Threats, Strengths, Synergies and Timing

- Views and comments are my own
- This audience is already familiar with much of this material – <u>but it does not appear in the national</u> <u>dialogue</u>
- A/J techniques and Data are from open literature
- Signal Powers (etc) are numbers for illustration and

comp

Brailferd ParMayn be +/- a few dB

Professor Emeritus (Recalled) Stanford University Toughening for PNTAB Dr. B. Parkinson

## <u>Attention Step:</u> Former High-Ranking DoD Official - A Visionary or ?

"I think that 20 years from now we won't be buying GPS satellites," he asserted. 'Twenty years from now everything you have that is manufactured for you, including your phone, will have, on the chip, a clock, a gyro and an accelerometer. It'll be set the moment it's manufactured and henceforth it will forever know what time it is, where it is, what its spatial orientation is. And it will never need a satellite.

## Headlines and Responses

- Press Headlines: <u>GPS vulnerable!</u>
  - Jamming
  - Spoofing
  - FCC authorization Blunders
- USG response Pursuit of <u>Augmentations</u>: "We have to find a replacement/backup"
  - A reasonable activity Studied for over 20 years (FAA-DME)

#### But, Current PNTAB Assessment:

"No <u>current</u> or <u>foreseeable</u> alternative to GNSS (Primarily GPS) can deliver equivalent accuracy (to millimeters, 3D) and world wide 24/7 availability."

#### It is time to <u>increase</u> the emphasis on well established solutions to ensure GNSS-based PNT.

12/9/2021

#### Background: Deliberate Jammer Alternatives

(Credit: Uncl. NATO Paper: Navigation Sensors and Systems in GNSS Degraded and Denied Environments)



Toughening for PNTAB

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Rockwell Collins GDM (1978) One of the Phase One User Sets (used over 10 kW of power)

- Apparent to me in 1973 that signal s to Jamming was an important issue
- We sponsored and encouraged AFAL Hi-A/J receiver with concerction from (JPO)
   Demonstrated over 100 dB of J/S!



- Field test Showed that a Hi-A/J GPS receiver could fly directly over a 10 KW jammer with no effect
- Result was forgotten for at least 20 years...

Repeating my Point: Much of what I have to say has been known and verified for over 40 Years – I think we need to balance the search for "Replacements" with a vigorous pursuit of <u>Toughening</u>

## <u>Historical Review :</u>

## A single Decibel (dB) = Ratio of 1.26

- Logarithmic ratio scale
  - dB is  $1/10^{\text{th}}$  of a Bell (which is a multiple of 10)
  - So  $10^{1/10}$  = 1.26. and 1.26<sup>10</sup> = 10.
- Definition originated in measurement of transmission loss and power in <u>telephony</u> (early 20th century) in the <u>Bell System</u>
- Named in honor of <u>Alexander Graham Bell</u>, (but Bel is seldom used.) Instead, dB used in science and <u>engineering</u>:
  - prominently in <u>acoustics</u>, <u>electronics</u>, and <u>control theory</u>.
  - Electronics, the <u>gains</u> of amplifiers, <u>attenuation</u> of signals, and <u>signal-to-noise ratios</u>

I will use dB - Jamming to Signal Power - as the <u>fundamental measure</u> of receiver effectiveness assuming a nominal L1C signal Power of -157 dBW <u>But:</u> I will use that J/S value to calculate the Jamming/Denial range of the selected (hypothesized) 1 Kw Ominidirectional Jammer

#### Capabilities of State-of-art GPS receivers with no Augmentations

Full Accuracy - State 5. Reduced Accuracy State 3

1.4 * 10 <sup>-16</sup>	in Received	<u>State 5</u> Data	State 3 Track				
- Watts	ower GPS III	Tolerable J/S	Tolerable J/S				
	(dBW)	(dB)	(dB)				
C/A	-158.5	34.0	44.7				
L1C	-157.0	35.7	52.7				
L2C	-158.5	39.2	47.7				
L5	-154.0	45.6	57.1				
C/A L1C L2C L5	-158.5 -158.5 -157.0 -158.5 -154.0	(ab) 34.0 35.7 39.2 45.6	(ab) 44.7 52.7 47.7 57.1				

State 5 = Code track, carrier track, data demodulation

State 3 = Code track only

Aside: Note that a Jammer's <u>denial area</u> for <mark>L5</mark> Full accuracy tracking is <mark>93% less</mark> than for L1 C/A Full tracking accuracy Translation from J/S (dB) to Maximum Jammer-Denial Range



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#### Denial Areas 1 Kw Jammer Located at Dallas Airport for unaugmented GPS L1C receiver State 5 Full Accuracy

(Assuming Line-of sight)



#### Scenario and Score Card

- Consider a Commercial Aircraft with full RTK accuracy (Code 5 tracking)
- Approaching and Landing at Dallas Fort Worth (DFW) (DFW has more than 15 GPS Approaches!)
- "Domesticated", 1 kW Jammer

(Reciver Data from Aerospace Corp. (Tom Powell & Phil Dafesh) for L1C signal and capability)

- "Raw" L1C Receiver-
  - Full Accuracy (state 5) signal Track J/S = 36 dB,
  - State 3 Code Tracking Accuracy J/S = 53 dB
- Minimum Satellite Earth Coverage power (L1C) = -157 dBW\*







#### The Goal to limit Jammer-Denial range to 500 meters

Our Goal: Use Receiver enhancements to reduce effective envelope of 1 kW jammer against L1C receiver to less than 500 meters



Jammer envelope if L1C Receiver Total J/S = 98 dB. [required enhancement of 60 dB for Full accuracy (State 5) tracking or 45 dB for reduced accuracy (State 3).] Achieving 62 dB of improvement in J/S Well-known Techniques for "Toughening" an L1 C/A receiver

- Category 1: Signal Processing Signal Modulation (L1 C/A or L1C) Tracking mode (State 5 - full accuracy or State 3 reduced accuracy) With or without "Vector Processing" <u>Category 2: Inertial Meas. + Low-phase noise User Clocks</u> MEMS - up to hi-grade IMU - Quartz to CSAC clocks '(Low Phase Noise) in user receivers **<u>Category 3: Controlled Reception Pattern Antennas</u> (CRPAs)**  Elements/Footprint - (4, 7 (Many) Beam/Null steering or combinations Category 4: Satellite Enhancements
  - Additional/Alternative Signals (Galileo, GLONASS (?), BeiDou (?))
  - Additional Frequencies (L5,) 2, Galileo)

## "Toughening" – nibbles and upgrades: <u>Category 1: Signal Processing</u>

		Range of improvement			Estimated
Technique		Low	High	Example	Time to Field
Receiver Techniques	L1C Code tracking (State 3)	10 dB	17 dB	17 dB	When L1C Operational
	Aircraft Shading	2 dB	4 dB	0 dB	Now
	Vector Receiver	4 dB	6dB	0 dB	Now to 5 yrs
	Totals – Signals and Processing	16 dB	27 dB	17 dB	Now to 5 years

Note: Modern receivers automatically revert to State 5 (Code Tracking) with the implication of reduced accuracy. I have made that step a part of "nibbles" <u>Takeaway</u>

These nibbles could produce a useful 10 to 27 dB of improvements against Goal of 62 dB improvement *Example will assume this "nibble" Category is* <u>17 dB</u>

This will be categorized as "Reduced Accuracy" or Code Tracking

Save the Aircraft shading and vector receiver for "Margin" against our goal

#### Using Code Tracking to reduce Max Jammer Denial Range





#### 1 Kw Jammer at Dallas Airport <u>denial areas</u> for GPS L1C receiver Effect of switching to Code Tracking (State 3) (Assuming Line-of sight)



## <u>Category 2</u> Nibbles: <u>Inertial</u> Synergies -

Well-Known Benefits

- Supports Longer Averaging Time for GPS/RF signal Best with "Tight-Coupling"
- Provides "Fly-wheeling" through outages
  - GPS to <u>calibrate</u> inertial components during valid reception periods
- Enable powerful <u>spoofing detection</u> and mitigation techniques - e.g.:
  - Velocity Verification
  - Enhances dual antenna heading verification
- If equipped with directional (beam) antenna: Provides accurate orientation measurements to enable precise beam steering during vehicle maneuvers

#### "Toughening" - nibbles and upgrades -<u>Category 2</u> <u>Inertial Synergies</u>

	Technique	Range of improvement			Estimated
		Low	High	Example	Time to Field
Receiver Enhance ment	Inertial & Averag. (MEMS, CSAC)	8 dB	20 dB	15 dB	Now

#### Takeaway

An unclassified Draper Paper, written for NATO, suggests Inertial synergies could improve J/S as much as 20 dB.

#### Will use 15 dB in our example





## Further Observations regarding Inertial Measurement Systems

- I advocate inertials but Inertial fly-wheeling is limited in accuracy:
  - Inertials are inherently vertically-unstable
  - Accelerometers do not measure acceleration
  - "Down" does not exactly point to center of the Earth - and locally deviates from models
  - "g" is <u>not just gravity</u>

So: Errors grow in Proportion to Time or Time<sup>2</sup>

## Elaboration -

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## The simple view of Inertial Navigation

• Double integrate <u>vector acceleration</u> and you have <u>vector position (i.e. 3D)</u>

$$\vec{P} = \int \int \vec{a} \ d^2 t$$

So with a perfect "accelerometer" you end up with perfect position??...
 <u>Absolutely not</u> -

#### "Perfect" accelerometers: What does an "Accelerometer" actually measure?



#### Gravity changes with altitude above the earth



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## The vertical dimension is *inherently* exponentially unstable



## The gravity vector -"Down" is only Local



## The user has to know the Initial Position and Velocity

• So we have:

 $\vec{P} = \int \int \vec{a} \ d^2t + \vec{V_0}t + \vec{P_0}$ <u>Current position</u> is known no better than
Initial position and the error increases with
time if initial velocity is not perfectly
known---

Where does an Inertial Measurement Unit find initial position?

# Another complication for inertial components

- To Navigate system must be accurately oriented to a known reference frame
- This converts the physical vectors to measurements that orient to E, N, and Up (or equivalent

• 
$$\begin{bmatrix} P_E \\ P_N \\ P_U \end{bmatrix} = \underline{P} = \int \int \left( \underline{f} + \underline{g} \right) d^2 t + \underline{V}_0 t + \underline{P}_0$$

 Note vector arrows have been replaced with underlines (indicating a coordinate system)

Wrap-up: Even Perfect "Accelerometers" can  
only be perfect non-field force sensors: They  
sense 
$$\vec{f} = \vec{a} - \vec{g}$$
 not  $\vec{a}$   
Thus total accel.:  $(\vec{a} = \vec{f} + \vec{g})$ 

- So PNT system has to accurately both Measure  $\vec{f}$  and calculate ...  $\vec{g}$
- Initial Alignment errors within "local" coordinate frame propagates errors
- Inertials are unstable sensors of <u>altitude</u> i.e. 2 Dimensional only

For fully robust receivers, all Inertial Systems benefit enormously with GNSS synergy

## Summary: Hi-Performance Inertial Navigator without GNSS (error growth at 0.3 nm/hour)



#### Synergy – GPS and <u>Tightly-coupled</u> Inertial (Regains GPS accuracy after 5 minute outage)



## Former High-Ranking DoD Official – A Visionary or ?



ears from now we won't be ," he asserted. 'Twenty years /thing you have that is pu, including your phone, will p a clock, a gyro and an

accelerometer. It'll be set the moment it's manufactured and henceforth it will forever know what time it is, where it is, what its spatial orientation is. And it will never need a satellite."

## **Category 3** Nibbles:

## Digital Beam and Null steering antennas

- GPS CRPAs well known for >40 years
  - Incorporated In Early, JPO Demo (1974 to 1978)
  - Many Journal Articles
  - <u>Internationally</u> well understood
  - Digital components readily available
  - Many manufacturers have developed and are selling CRPAs
- ITAR has limits on # of Elements in exported Receivers
  - Chinese and Russians probably do not adhere...
- Great striving to make small footprint but...
  - Hi-value (e.g. military vehicles/civilian Aircraft/ Maritime/longhaul trucks) mostly have both vehicle real estate and power
- Cost should greatly decrease with continuing advances in digital electronics, and large-scale use,

#### and with equipage when vehicle is manufactured

One Caution: Because the beam is formed with variable phase delays, both Code and Carrier tracking receivers must calibrate and account for this. JPALS program has successfully demonstrated the calibration techniques.

Basic Concept of Phased Array Φ

12/9/2021

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## Digital Antenna results from Bartone and Stansell public paper



Figure 1: 127-element L=Band CRPA Configuration

- Authors studied many configurations - up to 127 inexpensive antennas
- Analyzed the # versus performance tradeoff
- Currently prohibited by ITAR for greater than 4 elements for civil use

# Comparing elements and footprints for various CRPA configurations

CRPA Configurations with Approximate Dimensions														
		Number of Elements in Each Ring				Ring					Mounting Ring	Diameter (D)		
Rings	CE	1	2	3	4	5	6	Total # Elements	Directivity max [dB]	NB Signals Mitigated	r_i base on l_L1/2 [m]	Allocation [m]	D [m]	D [in]
0	1							1	2.0	0	0.000	0.095	0.10	3.75
1	1	6						7	14.5	6	0.095	0.170	0.36	14.19
2	1	6	12					19	.5.0	18	0.190	0.170	0.55	.1.5.
3	1	6	12	18				37	21.9	36	0.286	0.170	0.74	29.18
4	1	6	12	18	24			61	24.0	60	0.381	0.170	0.93	36.68
5	1	6	12	18	24	30		91	25.8	90	0.476	0.170	1.12	44.18
6	1	6	12	18	24	30	36	127	27.5	126	0.571	0.170	1.31	51.00

#### Feasibility with <u>off-the-shelf</u>

#### componenets:

"This antenna array can grow quite large in groundbased radar systems, with <u>over 100,000 elements</u>

being possible."

Data Sheet for 330 MHz 16 Bit A to D

> Price: about \$150 each



**INFUT FREQUENCY MAD** 

Toughening for PN TAB

340 350 358 379

Data Sheet

200 210 230 300 379 200 470

PRINCIPACY (NEW)

Figure 46. IF Pass Band Flatoess (locksdes Digital Filter)

NOMINAL PERFORMANCE FOR IF = 350 MHZ AND BW = 160 MHZ

shuffler enabled (every clock cycle), with default threshold settings, unless otherwise noted.

For = 350 MHz, BW = 160 MHz, Forc = 3.2 GHz, attenuator = 0 dB, Loc = 10 nH, maximum PIN\_0dBFS setting, form, pg = 266.7 MSP5,

# Four 49. Middlard Fragercy Approx Birber Digital Fibre





Figure 50. NED vs. CW input Power, CW at 155 MHz (NSD-Measured at 150 MHz at well at 150 MHz and 480 MHz/Band Edges)



#### AD6676

12/9/2021



#### Multiple Element Comparison Large Element Arrays can easily create multiple adaptive nulls



#### Signal to Interference Noise Ratio for large element, 1.2 meter Antenna



Figure 12: SINR Values for the 127-element CRPA with 5 Interference/Jammer Sources

- <u>With Five sources</u> of horizontal Interference
- Everywhere, at least 30 dB of Signal to (Interference plus noise) Ratio -or SINR

#### "Toughening" – nibbles and upgrades – <u>Category 3</u> Digital Beam Forming and Null steering

Tech		Range	of imp	rovement	Fatimated Time to Field	
lech	cnnique	Low	High	Example	Estimated Time to Field	
Dig Be Forr and N Ante	ital am ning Iulling enna	20 dB	45 dB	30 dB	Now to 5 Yrs	

<u> Takeaways:</u>

reduce

- At least 30 dB of improved  $J/S_0$  has been verified with hardware
- For good results, need about a 1-meter diameter Footprint
- Payoff exceeds penalty of finding space for certain users
- Also should enable enhanced situational awareness re: Jammers
- <u>At a median 30 dB improvement, this "nibble" alone can</u>

#### <u>Jammeroeffective area by 99.9%</u>





#### A quick summary to this point -Improvements to the <u>Receiver</u> System

Improvement Group	Median Improvement					
Signals and Processing	17 dB					
Tightly coupled Inertial	15 dB					
Digital Null and Beam Steering Antenna	30 dB					
Total Receiver Enhancements62 dB						
For Code Tracking L1C receiver, 36 + 62, or: J/S = 98						

- This has been roughly verified with real hardware
- All of these Nibbles should be <u>achievable in available users</u> <u>sets within 5 years</u>
- Impact on defeating <u>1KW</u>
  jammer:
  - Denial Slant Range Reduced from 556 Km to 0.4 Km
  - Area of Denial Jamming reduced from 972,000 Km<sup>2</sup> to 0.6 Km<sup>2</sup>.



Toughening for PNTAB Dr. B. Parkinson "Toughening" - nibbles and upgrades <u>Category 4</u>: Satellite

#### Enhancements

- Additional/Alternative Signals (Galileo, GLONASS (?), Beidou(?))
- Additional Frequencies (L5, L2, Galileo)

#### Comparing L5 and L1C based on Max Jammer-Denial Range Note: FAA is pursuing L5, but apparently not L1C



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# Spoofing



#### Additional observations re: spoofing-Additional Detection Techniques

- <u>Directional Antennas</u> can attenuate spoofing as well as reduce noise jammer interference
  - Amplify Valid Signal
  - Attenuate Spoofing input
  - Measure bearing of Spoofer
- **<u>RF environment monitoring</u>**: local, regional, national
  - Input Power above normal
- External Detection and Notification FAA's WAAS? (J911)
- Other System Crosschecks
  - Inertial Navigation Components
  - Other RF Systems LEOs, eLoran or FAA's DME
  - Eyeballs / Magnetic Compass etc.

#### Summary reminder: The Jammer Threat is real and growing

Chinese - engineers from Tsinghua University in Beijing



Also: https://ctstechnologys.com/low-altitude-gps-spoofing-system-drone-defense-anti-drones-device.html

## Summary and Conclusions

"GNSS (GPS) cannot be matched with any terrestrial system in terms of accuracy, 3D, Worldwide 24/7, but must be protected against Jamming"

- The civil jammer threat is very real and rapidly growing. Aviation, including RPVs are particularly threatened. Maritime is also very vulnerable.
- More emphasis should be placed on <u>toughening</u> GPS against high-powered Jammers:
  Extreme resilience can be created with a modern Receiver <u>System</u>
  - The most important contribution Category is a Multi-element (>18) Digital Beam Forming and Null steering Antennas
  - These techniques are also powerful anti-spoofing tools
  - New improvements should be ready to field in the next 3 to 5 years if deemed urgent
  - Many companies are actively pursuing these techniques
- While Inertial Systems can flywheel through GPS outages, they must be periodically reset because of unbounded error growth -<u>even with perfect "accelerometers"</u>
- FAA can help by emphasizing Toughened GNSS Receivers, particularly using directional antennas.
  - ITAR antenna restrictions must be removed. They are only hurting the US, since the whole world knows the technique and has access to the commercial components.
- The L1C and L5 signals are much more robust than the L1 C/A. The FAA should rapidly enable aviation to use these signals, including WAAS, enhanced GPS, and the supporting MOPS etc.

# A recommendation

- That PNTAB forms a committee on "Toughening" focusing on, at least, countering both jamming and <u>spoofing</u>
  - Identify and project the civil threats
  - Identify mitigations and roadblocks to implementations
  - Create a report and recommendations to the EXCOM for USG actions
- Members?
  - Tom Powell, John Betz, Frank Van Diggelen, Scott Burgett, et.al.
  - Advisors: Chris Hegarty, Ken Alexander, Karen Van Dyke

Let's re-emphasize "Toughening" and develop affordable multielement antennas.

And remove them from the Munitions List so Commercial airplanes can exploit, and the COTS prices drop.



# Backups







### Three Action Areas:

<u>PTA</u> - <u>P</u>rotect, <u>Toughen</u>, <u>A</u>ugment

- Joughen <u>Users'</u> <u>Receivers</u> to use GNSS
  - Employ multiple, well known techniques to ensure <u>spoofing</u> can never create HMI
  - Increase Jam resistance use well established techniques
  - Diversify <u>All integrity-certified GNSS signals</u> receivers (with vector feature)

Assured Availability of PNT - the "PTA" Strategy

#### **Three Action Areas**:

<u>PTA</u> - <u>P</u>rotect, <u>T</u>oughen, <u>Augment</u>



# Deliberate Spoofing has been Demo "Professor fools \$80M superyacht's GPS receiver on

#### Many examples of Spoofing recently, Real and Possible:

- Academic Demonstrations
- Possible Incidents for Military
- Will focus on "Civilian" Receivers
- Military has additional anti-• spoofing techniques



Ionian Sea in late June 2013 and early

- Outline:
  - What is Spoofing? "White Rose of Drachs" yacht captain.
  - How can it be prevented?
  - What actions might USG take?

July 2013 with the full consent of the

Spoofing Definition and General Techniques

#### <u>Spoofing:</u>

 Deliberately creating False GNSS signals that lead to misleading Position, Time or Velocity

Note: Not considering <u>inadvertent satellite errors</u> –an integrity problem, albeit has some of the same solutions

• <u>A Few</u> Examples of Deliberate Spoofing Techniques

<u>Technique 1</u>. <u>Create</u> fictitious signals & broadcast to user

- Presumably Hazardous and Misleading Information ("HMI")
- Requires Knowledge of Signal Sequences
- Requires time synchronization

<u>Technique 2</u>. <u>Rebroadcast</u> GPS signals with >> Power

Arrives at user with delay – nanosecs to 10s of microseconds <u>Technique 3</u>. Combination of 1 and 2.









# Spoofing Summary

- "Competent" (Skeptical) receivers should detect spoofing
  - At a minimum, cleanly stop providing misleading outputs
  - Consistency checking ("crosschecking" a self-integrity monitor)
  - Many other techniques e.g. directional antennas
- Many Receivers should be able to "Operate Through"

Well-known defenses are beginning to be incorporated

# Finding Initial Attitude for INS

- Null two cross axis accelerometers to find "level"
- Orient East/West gyro to sense no earth rate
- Typically takes 15 to 20 minutes to find orientation to about an arc minute
- At 100km, an arc minute in azimuth is about 30 meters.
- <u>Note</u>: With GPS aiding, initial alignment can occur in the first 30 minutes of flight with no waiting on the flight line.

#### L1C A/J techniques against 1 kW

Tammer								
	Full Kin	State 5 ematic A	ccuracy	State 3 Code Tracking Accuracy				
	dB of J/S	Range (km)	Area (km²)	dB of J/S	Range (km)	Area (km²)		
No AJ	36	556	972,000	53	79	19,390		
+ Inertial only	51	99	30,731	68	14	613		
+ Dir. Antenna only	66	18	971	83	2.5	19		
+ Inert. & Dir. Antenna	81	3.1 Toughe	<b>31</b> hing for PNTAB	98	0.4	0.6		

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#### <u>L5</u> A/J techniques against 1 kW Jammer

	Full Kin	State 5 ematic A	ccuracy	State 3 Code Tracking Accuracy			
	dB of J/S	Range (km)	Area (km²)	dB of J/S	Range (km)	Area (km²)	
No AJ	46	120	45,454	57	34	3610	
+ Inertial only	61	21	1437	68	9.6	287	
+ Dir. Antenna only	76	3.8	45	83	1.7	9.1	
+ Inert. & Dir. Antenna	91	<b>0.7</b> Tougher Dr.	<b>1.4</b> hing for PNTAB B. Parkinson	102	0.2	0.1 72	
## <u>GPS L1C Receiver</u>. Maximum Radius (Km) of 1K Jammer for Various A/J capabilities -



## <u>GPS L5 Receiver</u>. Maximum Radius (Km) of 1K Jammer for Various A/J capabilities -





## L1C and L5 A/J Comparisons Max Radius of 1 Kw Jammer

