#### Change Topic: Public Signals-in-Space (SiS) Updates

This change package accommodates the text changes to support the proposed solution (see table below) within the public Signals-in-Space (SiS) documents. All comments must be submitted in Comments Resolution Matrix (CRM) form.

The columns in the WAS/IS table following this page are defined below:

Section Number: This number indicates the location of the text change within the document.

Proposed Heading: Contains existing and/or proposed changes to section titles and/or the titles to new sections

(WAS) <Document Title>: Contains the baseline text of the impacted document.

Proposed Object Text: Contains proposed changes to baseline text.

Proposed Rationale: Contains the supporting information to explain the reason for the proposed changes.

#### PROBLEM STATEMENT:

There are eight areas of obsolete/ambiguous language in the Signals-in-Space (SiS) specifications (mean anomaly equation, convolutional encoding, LNAV special messages reference, Universal Coordinated Time Offset Error (UTCOE), User Range Accuracy (URA) Note #3, Right Ascension Angle Language, and the signal health versus navigation data terminology, publication errors). If this language were interpreted incorrectly it could result in UE developers designing receivers that don't work.

**SOLUTION:** (Proposed)

Resolve the obsolete/ambiguous language in the areas above to avoid the potential for misinterpretation.

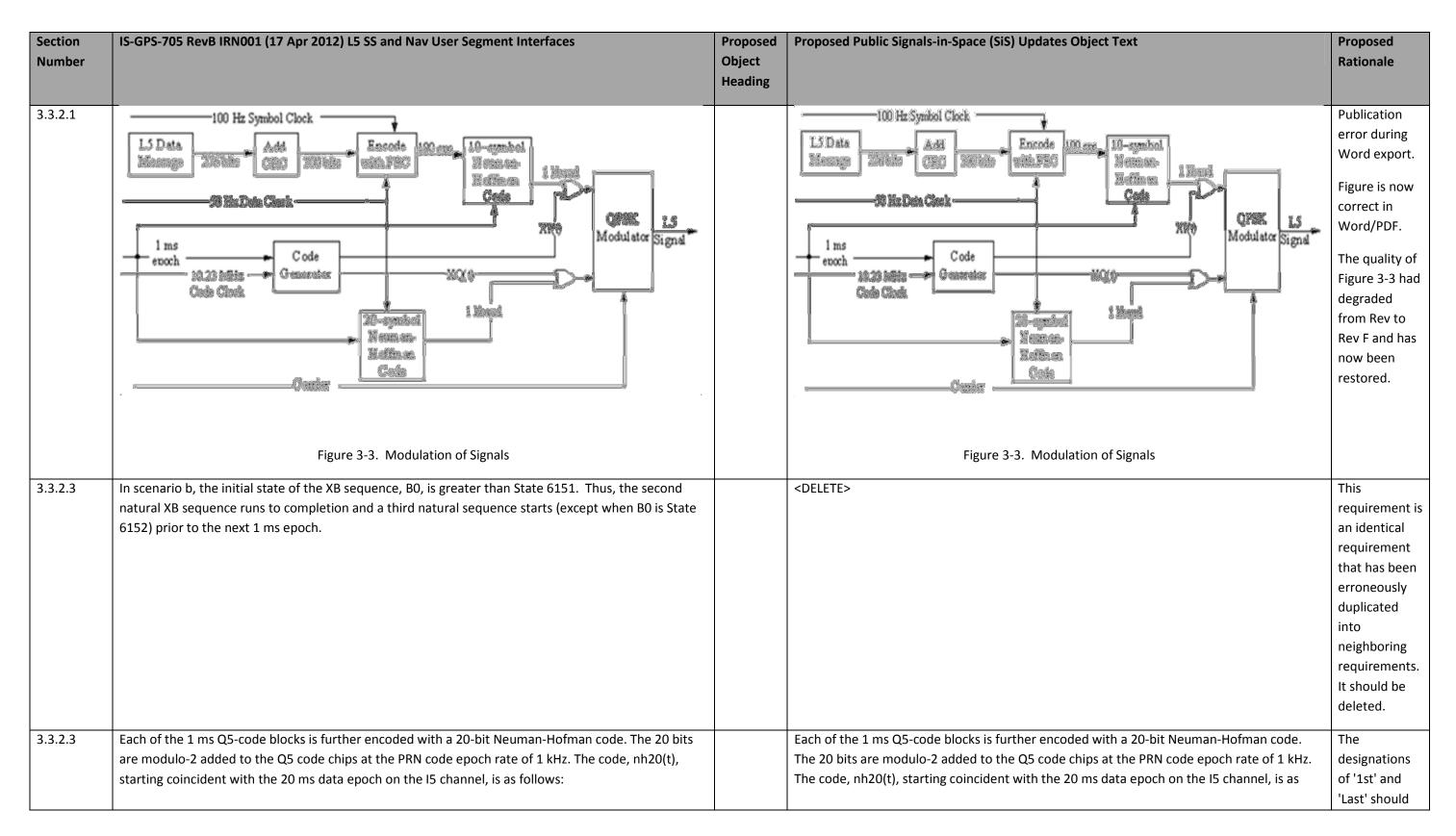
Note: For the changes with respect to IS-GPS-705B, IRN-001 there are *four* areas that are being amended:

- i. Coordinated Universal Coordinated Time Offset Error (UTCOE), (1 proposed change)
- ii. Signal health versus navigation data terminology) (1 proposed change)
- iii. Mean Anomaly equation (1 proposed change)
- iv. Publication Errors (32 proposed changes)

#### Start of WAS/IS for IS-GPS-705B, IRN-001 Changes

Section Number	IS-GPS-705 RevB IRN	N001 (17 Apr 2	2012) L5 SS and Nav User	Segment Interfaces	Proposed Object Heading	Proposed Public Signals-in-Sp	ace (SiS) Updates Objec	t Text	Proposed Rationale
2.2	Other Publications International Earth R	Rotation and R	eference Systems Service	e (IERS) Technical Note 36		<u>Other Publications</u> International Earth Rotation a	nd Reference Systems S	ervice (IERS) Technical Note 36	In order to be consistent with previous Revision, the term "Other Publications" should be underlined.
3.3.1.6		Table 3-	-III. Received Minimum RF Sig	gnal Strength		Table	3-III. Received Minimum RF S		Updated terminology
			S	Signal		SV		Signal	to "SV", "Bloc IIF" and "GPS
	SV	V Blocks	15	Q5			15	Q5	IIF" and "GPS III" for
		IIF	-157.9 dBW	-157.9 dBW		Block IIF	-157.9 dBW	-157.9 dBW	consistency.
		Ш	-157.0 dBW	-157.0 dBW		GPS III	-157.0 dBW	-157.0 dBW	
3.3.1.6.1		and Subsequent		Minimum RF Signal Strength Specified in 3.3.1.1 – GEO Bas		Table 3-IV.       Space Se         GPS III and Subsequer         Antennas	t Satellites over the Bandwidt	d Minimum RF Signal Strength for h Specified in 3.3.1.1 – GEO Based	Updated terminology to "SV", and
	CVI	D11	S	ignal		SV		Signal	"GPS III and
	501	Blocks	I5	Q5			15	Q5	Subsequent Blocks" for
	III and S	Subsequent	-182.0 dBW	-182.0 dBW		GPS III and Subsequent Blocks	-182.0 dBW	-182.0 dBW	consistency.
		I							

UNCLASSIFIED Change Topic: Public Signals-in-Space (SiS) Updates



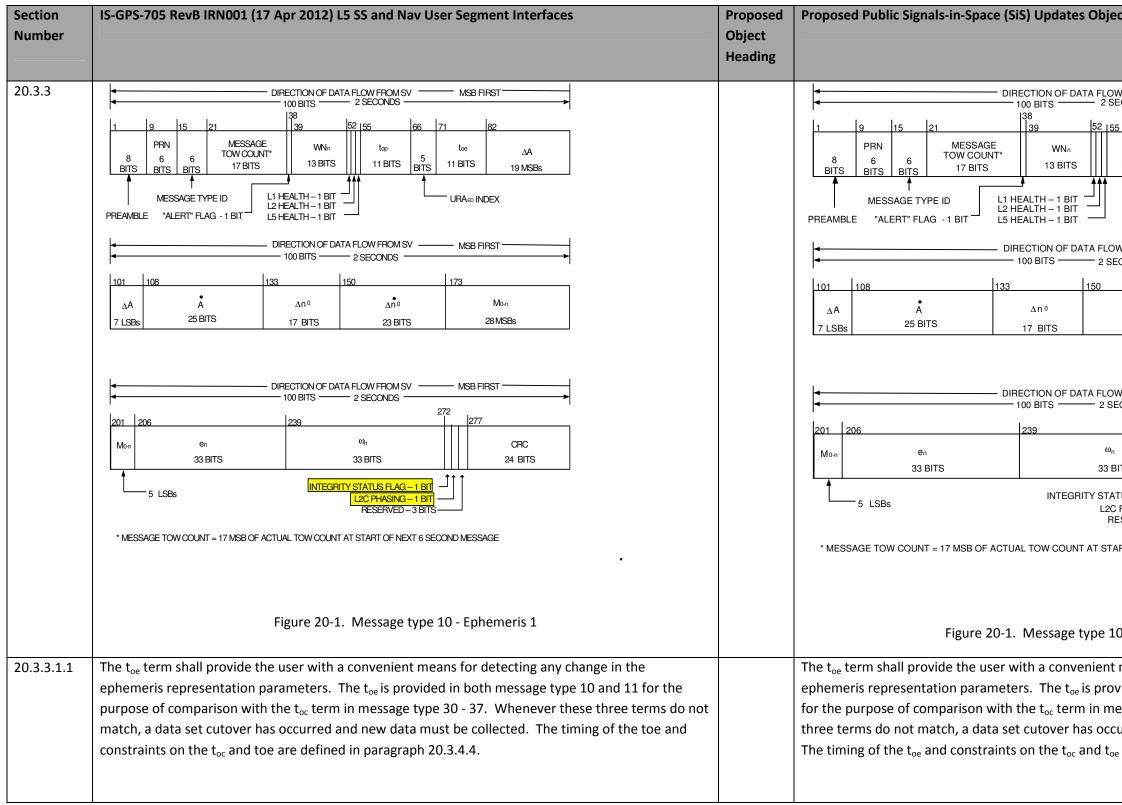
Section Number	IS-GPS-705 RevB IRN001 (17 Apr 2012) L5 SS and Nav User Segment Interfaces	Proposed Object Heading	Proposed Public Signals-in-Space (SiS) Updates Object Text	Proposed Rationale
	1 <sup>st</sup> Last nh20(t) = 0 0 0 0 0 1 0 0 1 1 0 1 0 1 0 1 1 1 0		follows: $1^{st}$ Last $nh20(t) = 0\ 0\ 0\ 0\ 1\ 0\ 0\ 1\ 1\ 0\ 1\ 0\ 0\ 1\ 1\ 0$	be aligned properly to the bit stream.
3.3.4	The L5 CNAV data contains the requisite data for relating GPS time to UTC. The accuracy of this data during the transmission interval will be such that it relates GPS time to UTC (USNO) to within 90.0 nanoseconds (one sigma). This data is generated by the CS (or provided to the CS); therefore, the accuracy of these relationships may degrade if for some reason the CS is unable to upload data to an SV.		The L5 CNAV data contains the requisite data for relating GPS time to UTC. This data is generated by the CS (or provided to the CS); therefore, the accuracy of these relationships may degrade if for some reason the CS is unable to upload data to an SV.	The text "The accuracy of this data during the transmission interval shall be such that it relates GPS time (maintained by the MCS of the CS) to UTC (USNO) within 90 nanoseconds (one sigma)" has been deleted. The rationale is that the time accuracy stated (90ns- one sigma) is not aligned to the SPS PS (40ns).

Section Number	IS-GPS-705 RevB IRN001 (17 Apr 2012) L5 SS and Nav User Segment Interfaces	Proposed Object Heading	Proposed Public Signals-in-Space (SiS) Updates Object Text	Proposed Rationale
6.2.2.2.1	Block IIA SVs	Block II SVs		Inserting Section 6.2.2.2.1 Block II SVs to maintain synchronizatio n with IS-GPS- 200.
6.2.2.2.1			See paragraph 6.2.2.2.1 of IS-GPS-200. These satellites do not broadcast the L5 signal.	Inserting Section 6.2.2.2.1 Block II SVs to maintain synchronizatio n with IS-GPS- 200.
6.2.2.2	Block IIR SVs.	Block IIA SVs.		Section 6.2.2.2.1 Block IIA SVs moved to Section 6.2.2.2.2. All associated text with 6.2.2.2.1 Block IIA SVs has also moved to Section 6.2.2.2.2.
6.2.2.2.3	Block IIR-M SVs.	Block IIR SVs.		Section 6.2.2.2.33Bloc k IIR- SVs moved to Section

Section Number	IS-GPS-705 RevB IRN001 (17 Apr 2012) L5 SS and Nav User Segment Interfaces	Proposed Object Heading	Proposed Public Signals-in-Space (SiS) Updates Objec
6.2.2.2.4			
6.2.2.2.4	<u>Block IIF SVs</u> .	<u>Block IIR-</u> <u>M SVs.</u>	
6.2.2.2.5	<u>GPS III SVs</u>	<u>Block IIF</u> <u>SVs</u> .	

oject Text	Proposed
	Rationale
	6.2.2.3.
	All associated
	text with
	6.2.2.2.2 Block
	IIR SVs has
	also moved to
	Section
	6.2.2.3.
	Section
	6.2.2.2.3 Block
	IIR-M SVs
	moved to
	Section
	6.2.2.2.4.
	All associated
	text with
	6.2.2.2.3 Block
	IIR M SVs has
	also moved to
	Section
	6.2.2.2.4.
	Section
	6.2.2.2.4 Block
	IIF SVs moved
	to Section
	6.2.2.2.5.
	All associated
	text with
	6.2.2.2.4 Block
	IIR M SVs has
	also moved to
	Section

6.2.2.2.6				Rationale
6 9 9 9 6				6.2.2.2.5.
6.2.2.2.6		GPS III		Section
		SVs		6.2.2.2.6 GPS
				III SVs moved
				from Section
				6.2.2.2.5 to
				remain
				synchronized
				with IS-GPS-
				200.
6.4.4	PRNs 33 and 63	PRNs 33		Fixed typo
		through		from "33 and
		63		63" to "33
				through 64"
10.3	Any letter of exception which is in force for the revision of the IS is depicted in Figure 10.3-1, 1-0.3-2,		Any letter of exception which is in force for the revision of the IS is depicted in Figure 10.3-1,	Fixed figure
	and 10.3-3.		10.3-2, 10.3-3, and 10.3-4.	references.
				Also included
				reference to
				10.3-4.



ct Text		Proposed Rationale
W FROM SV	toe ΔA 11 BITS 19 MSBs URA €D INDEX - MSB FIRST	Removing yellow highlighting from Figure 20-1. The highlighted bits were only differentiated for discussion purposes and removing the highlights have no technical impact.
vided in both m essage type 30 urred and new	ecting any change in the essage type 10 and 11 - 37. Whenever these data must be collected. paragraph 20.3.4.4.	The 't <sub>oe</sub> ' term in the last sentence should be subscripted to remain consistent

Section Number	IS-GPS-705 RevB IRN001 (17 Apr 2012) L5 SS and Nav User Segment Interfaces	Proposed Object Heading	Proposed Public Signals-in-Space (SiS) Updates Object Text       Proposed Ration         Ration       Proposed Public Signals-in-Space (SiS) Updates Object Text	
			terms previo the pa and	the 't <sub>oe</sub> ' as listed iously in paragraph ughout IS- 705.
20.3.3.1.1	Any change in the message type 10 and 11 ephemeris data will be accomplished with a simultaneous change in the $t_{oe}$ value. The CS will assure the $t_{oe}$ value for Block IIR-M/IIF and SS will assure the toe value for GPS III, for at least the first data set transmitted by an SV after an upload, is different from that transmitted prior to the cutover. See Section 20.3.4.5 for additional information regarding $t_{oe}$ .		simultaneous change in the t <sub>oe</sub> value. The CS will assure the t <sub>oe</sub> value for Block IIR-M/IIF and SS will assure the t <sub>oe</sub> value for GPS III, for at least the first data set transmitted by an SV after an upload, is different from that transmitted prior to the cutover. See Section 20.3.4.5 for additional information regarding t <sub>oe</sub> .	IId be cripted to ain istent the 't <sub>oe</sub> ' ns listed iously in paragraph ughout IS-
20.3.3.1.1.4	URA <sub>ED</sub> IndexURA <sub>ED</sub> (meters)156144.00< URA <sub>ED</sub> (or no accuracy prediction is available)		the following relationship to the ED URA: senter	scriptive ence was
	15       6144.00 $<$ URA <sub>ED</sub> (or no accuracy prediction is available)         14       3072.00 $<$ URA <sub>ED</sub> $\leq$ 6144.00         13       1536.00 $<$ URA <sub>ED</sub> $\leq$ 3072.00		URA <sub>ED</sub> IndexURA <sub>ED</sub> (meters)droppe156144.00< URA <sub>ED</sub> (or no accuracy predictionthis reliand nois available) </td <td>neously ped from revision now has</td>	neously ped from revision now has
	$13 \qquad 1350.00 \qquad < OKA_{ED} \leq 3072.00$ $12 \qquad 768.00 < URA_{ED} \leq 1536.00$		143072.00 $<$ URA <sub>ED</sub> $\leq$ 6144.00been reinset	
	11 $384.00 < URA_{ED} \leq 768.00$		13 1536.00 < URA <sub>ED</sub> ≤ 3072.00	
			12 768.00 $< URA_{ED} \le 1536.00$	

Section Number	IS-GPS-705 RevB II	RN001 (17 Apr 2012) L5 SS and	l Nav User Segment Interfaces	Proposed Object Heading	Proposed Public Si	gnals-in-Space (SiS) Updates Object Text	Proposed Rationale
	10	192.00 < URA <sub>ED</sub> ≤	384.00		11	$384.00 < URA_{ED} \leq 768.00$	
	9	96.00 < URA <sub>ED</sub> $\leq$	192.00		10	$192.00 < URA_{ED} \leq 384.00$	
	8	$48.00 < \text{URA}_{\text{ED}} \leq$	96.00		9	96.00 < URA <sub>ED</sub> ≤ 192.00	
	7	$24.00 < \text{URA}_{\text{ED}} \leq$	48.00		8	$48.00 < URA_{ED} \leq 96.00$	
	6	$13.65 < URA_{ED} \leq$	24.00		7	$24.00 < URA_{ED} \le 48.00$	
	5	$9.65 < URA_{ED} ≤$	13.65		6	13.65 < URA <sub>ED</sub> ≤ 24.00	
	4	$6.85 < URA_{ED} \leq$	9.65		5	9.65 < URA <sub>ED</sub> ≤ 13.65	
	3	4.85 < URA <sub>ED</sub> ≤	6.85		4	6.85 < URA <sub>ED</sub> ≤ 9.65	
	2	$3.40 < URA_{ED} \leq$	4.85		3	$4.85 < URA_{ED} \le 6.85$	
	1	2.40 < URA <sub>ED</sub> ≤	3.40		2	$3.40 < URA_{ED} \le 4.85$	
	0	1.70 < URA <sub>ED</sub> ≤	2.40		1	$2.40 < URA_{ED} \leq 3.40$	
	-1	$1.20 < URA_{ED} \leq$	1.70		0	$1.70 < URA_{ED} \leq 2.40$	
	-2	$0.85 < URA_{ED} \leq$	1.20		-1	$1.20 < URA_{ED} \le 1.70$	
	-3	$0.60 < URA_{ED} \leq$	0.85		-2	0.85 < URA <sub>ED</sub> ≤ 1.20	
	-4	$0.43 < URA_{ED} \leq$	0.60		-3	0.60 < URA <sub>ED</sub> ≤ 0.85	
	-5	$0.30 < URA_{ED} \leq$	0.43		-4	$0.43 < URA_{ED} \le 0.60$	
	-6	$0.21 < URA_{ED} \leq$	0.30		-5	0.30 < URA <sub>ED</sub> ≤ 0.43	

Section Number	IS-GPS-705 RevB I	RN001 (17 Apr 2012) L5 SS and Nav User Segment Interfaces	Proposed Object Heading	Proposed Public S	ignals-in-Space (SiS) Updates Object Text	Proposed Rationale
	-7	$0.15 < URA_{ED} \leq 0.21$		-6	$0.21 < URA_{ED} \leq 0.30$	
	-8	$0.11 < URA_{ED} \leq 0.15$		-7	$0.15 < URA_{ED} \leq 0.21$	
	-9	$0.08 < URA_{ED} \leq 0.11$		-8	$0.11 < URA_{ED} \leq 0.15$	
	-10	$0.06 < URA_{ED} \le 0.08$		-9	$0.08 < URA_{ED} \leq 0.11$	
	-11	$0.04 < URA_{ED} \leq 0.06$		-10	$0.06 < URA_{ED} \leq 0.08$	
	-12	$0.03 < URA_{ED} \leq 0.04$		-11	$0.04 < URA_{ED} \leq 0.06$	
	-13	$0.02 < URA_{ED} \leq 0.03$		-12	$0.03 < URA_{ED} \leq 0.04$	
	-14	$0.01 < URA_{ED} \leq 0.02$		-13	$0.02 < URA_{ED} \leq 0.03$	
	-15	$URA_{ED} \leq 0.01$		-14	$0.01 < URA_{ED} \leq 0.02$	
	-16	No accuracy prediction available-use at own risk		-15	$URA_{ED} \leq 0.01$	
				-16	No accuracy prediction available-use at own risk	
				For each URA inde	ex (N), users may compute a nominal URA value (X) as given by:	
	For each URA inde	ex (N), users may compute a nominal URA value (X) as given by:		• If the va	lue of N is 6 or less, $X = 2^{(1 + N/2)}$ ,	
	• If the va	lue of N is 6 or less, $X = 2^{(1 + N/2)}$ ,		• If the va	lue of N is 6 or more, but less than 15, $X = 2^{(N-2)}$ ,	
	• If the va	lue of N is 6 or more, but less than 15, $X = 2^{(N-2)}$ ,			hall indicate the absence of an accuracy prediction and shall advise	the
	• N = 15 s	hall indicate the absence of an accuracy prediction and shall advise the standard			itioning service user to use that SV at his own risk.	
		ervice user to use that SV at his own risk.		For N = 1, 3, and 5	, X should be rounded to 2.8, 5.7, and 11.3 meters, respectively.	
	For N = 1, 3, and 5	, X should be rounded to 2.8, 5.7, and 11.3 meters, respectively.		The nominal URA-	$_{\rm D}$ value (X) is suitable for use as a conservative prediction of the RN	1S FD
	The nominal URA	<sub>p</sub> value (X) is suitable for use as a conservative prediction of the RMS ED range			ccuracy-related purposes in the pseudorange domain (e.g., measur	
		y-related purposes in the pseudorange domain (e.g., measurement de-weighting,		-	M, FOM computations). Integrity properties of the IAURA <sub>ED</sub> are spe	
		utations). Integrity properties of the IAURA <sub>ED</sub> are specified with respect to the scale	d	with respect to the	e scaled (multiplied by either 4.42 or 5.73 as appropriate) upper bo	ound

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	(multiplied by either 4.42 or 5.73 as appropriate) upper bound values of the broadcast URA <sub>ED</sub> index (see 20.3.3.1.1). For the nominal URA <sub>ED</sub> value and the IAURA <sub>ED</sub> value, users may compute an adjusted URA <sub>ED</sub> value as a function of SV elevation angle (E), for $E \ge 0$ , as follows:		values of the broadcast URA <sub>ED</sub> index (see 20.3.3.1.1). For the nominal URA <sub>ED</sub> value and the IAURA <sub>ED</sub> value, users may compute an adjusted URA <sub>ED</sub> value as a function of SV elevation angle (E), for $E \ge 0$ , as follows: Adjusted Nominal URA <sub>ED</sub> = Nominal URA <sub>ED</sub> (sin(E+90 degrees))	
	Adjusted Nominal URA <sub>ED</sub> = Nominal URA <sub>ED</sub> (sin(E+90 degrees)) Adjusted IAURA <sub>ED</sub> = IAURA <sub>ED</sub> (sin(E+90 degrees))		Adjusted IAURA <sub>ED</sub> = IAURA <sub>ED</sub> (sin(E+90 degrees)) URA <sub>ED</sub> and IAURA <sub>ED</sub> account for SIS contributions to user range error which include, but are not limited to, the following: LSB representation/truncation error, alongtrack ephemeris errors, and crosstrack ephemeris errors. URA <sub>ED</sub> and IAURA <sub>ED</sub> do not account for user range error contributions due to the inaccuracy of the broadcast ionospheric data parameters	
	URA <sub>ED</sub> and IAURA <sub>ED</sub> account for SIS contributions to user range error which include, but are not limited to, the following: LSB representation/truncation error, alongtrack ephemeris errors, and crosstrack ephemeris errors. URA <sub>ED</sub> and IAURA <sub>ED</sub> do not account for user range error contributions due to the inaccuracy of the broadcast ionospheric data parameters used in the single-frequency ionospheric model or for other atmospheric effects.		used in the single-frequency ionospheric model or for other atmospheric effects.	

S-GPS-705 F	RevB IRN001 (17 Apr 2012) L5 SS and Na	v User Se	gment In	terfaces		Proposed I Object Heading	Proposed P	ublic Signals-in-Space (SiS) Updates	Object 1	ext		
Table 20-I.     Message Types 10 and 11 Parameters (2 of 2)								Table 20-I. Message Types 10	and 11 Pa	rameters (2	2 of 2)	
Parameter Symbol	Parameter Description	No.of Bits**	Scale Factor (LSB)	Effective Range***	Units		Parameter Symbol	Parameter Description	No. of Bits**	Scale Factor (LSB)	Effective Range***	Units
t <sub>oe</sub>	Ephemeris data reference time of week	11	300	604,500	seconds		t <sub>oe</sub>	Ephemeris data reference time of week	11	300	604,500	seconds
Ω <sub>0-n</sub> ****	Reference right ascension angle	33*	2-32		semi-circles		$\Omega_{0-n}$	Longitude of Ascending Node of Orbit Plane at Weekly Epoch	33*	2 <sup>-32</sup>		semi-circles
$\Delta \Omega^{\bullet}$ *****	Rate of right ascension difference	17*	2 <sup>-44</sup>		semi-circles/sec		$\Delta \Omega^{\bullet}$ ****	Rate of right ascension difference	17*	2-44		semi-circles/sec
i <sub>0-n</sub>	Inclination angle at reference time	33*	2 <sup>-32</sup>		semi-circles		i <sub>0-n</sub>	Inclination angle at reference time	33*	2 <sup>-32</sup>		semi-circles
i₀•n	Rate of inclination angle	15*	2-44		semi-circles/sec		i₀-n	Rate of inclination angle	15*	2-44		semi-circles/sec
C <sub>is-n</sub>	Amplitude of the sine harmonic correction term to the angle of inclination	16*	2 <sup>-30</sup>		radians		C <sub>is-n</sub>	Amplitude of the sine harmonic correction term to the angle of inclination	16*	2 <sup>-30</sup>		radians
C <sub>ic-n</sub>	Amplitude of the cosine harmonic correction term to the angle of inclination	16*	2 <sup>-30</sup>		radians		C <sub>ic-n</sub>	Amplitude of the cosine harmonic correction term to the angle of inclination	16*	2 <sup>-30</sup>		radians
C <sub>rs-n</sub>	Amplitude of the sine correction term to the orbit radius	24*	2-8		meters		C <sub>rs-n</sub>	Amplitude of the sine correction term to the orbit radius	24*	2-8		meters
C <sub>rc-n</sub>	Amplitude of the cosine correction term to the orbit radius	24*	2-8		meters		C <sub>rc-n</sub>	Amplitude of the cosine correction term to the orbit radius	24*	2-8		meters
C <sub>us-n</sub>	Amplitude of the sine harmonic correction term to the argument of latitude	21*	2-30		radians		C <sub>us-n</sub>	Amplitude of the sine harmonic correction term to the argument of latitude	21*	2-30		radians
C <sub>uc-n</sub>	Amplitude of the sine harmonic correction term to the argument of latitude	21*	2-30		radians		C <sub>uc-n</sub>	Amplitude of the sine harmonic correction term to the argument of latitude	21*	2-30		radians
** Se *** U in **** Ω of	arameters so indicated are two's complement, we Figure 20-1 and Figure 20-2 for complete bit nless otherwise indicated in this column, eff dicated bit allocation and scale factor. $p_{eff}$ is the right ascension angle at the weekly epering fright ascension { $\hat{\Omega}_{REF}$ Table 20-II }. elative to $\hat{\Omega}_{REF} = -2.6 \times 10^{-9}$ semi-circles/second	allocation fective ranges och ( $\Omega_{0-w}$ )	in message ge is the 1	e types 10 and maximum rar	111; nge attainable with		** Se *** U in	arameters so indicated are two's complement, v ee Figure 20-1 and Figure 20-2 for complete bit inless otherwise indicated in this column, eff dicated bit allocation and scale factor. elative to $\hat{\Omega}_{\text{REF}} = -2.6 \times 10^{-9}$ semi-circles/second	allocation	in message	types 10 and	11;

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				Ascending
				Node of Orbit
				Plane at
				Weekly
				Epoch" in IS-
				GPS-200 Table
				20-II.
				Recommend
				that the
				identical
				terms be used since they are
				have identical
				definitions.
				definitions.
				Also
				recommendin
				g deleting the
				4-star note,
				and thus
				renumbering
				the old 5-star
				note to a 4-
				star note.
20.3.3.2.4	The CS shall derive URA <sub>NED0</sub> , URA <sub>NED1</sub> , and URA <sub>NED2</sub> indexes which, when used together in the above		The CS shall derive URA <sub>NED0</sub> , URA <sub>NED1</sub> , and URA <sub>NED2</sub> indexes that, when used together in the	Corrected the
	equations, results in the minimum IAURA <sub>NED</sub> that is greater than the predicted IAURA <sub>NED</sub> during		above equations, results in the minimum IAURA <sub>NED</sub> that is greater than the predicted	grammar typo
	the/ephemeris fit interval.		IAURA <sub>NED</sub> during the ephemeris fit interval.	by changing
				"which,
				results" to
				"that,result"
				and corrected
				the typo by
				changing
				"the/ephemer
				is" to "the

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				ephemeris".
20.3.3.2.4	For each URA <sub>NED0</sub> index (N), users may compute a nominal URA <sub>NED0</sub> value (X) as given by:		For each $URA_{NED0}$ index (N), users may compute a nominal $URA_{NED0}$ value (X) as given by:	Changing:
	• If the value of N is 6 or less, but more than -16, $X = 2^{(1 + N/2)}$ ,		• If the value of N is 6 or less, but more than -16, $X = 2^{(1 + N/2)}$ ,	
	• If the value of N is 6 or more, but less than 15, $X = 2^{(N-2)}$ ,		• If the value of N is 6 or more, but less than 15, $X = 2^{(N-2)}$ ,	"The
	• N = -16 or N = 15 shall indicate the absence of an accuracy prediction and shall advise the standard positioning service user to use that SV at his own risk.		• N = -16 or N = 15 shall indicate the absence of an accuracy prediction and shall advise the standard positioning service user to use that SV at his own risk.	transmitted URA <sub>NED2</sub> index is an integer
	For N = 1, 3, and 5, X should be rounded to 2.8, 5.7, and 11.3 meters, respectively.		For N = 1, 3, and 5, X should be rounded to 2.8, 5.7, and 11.3 meters, respectively.	value in the range 0 to 7.
	The nominal URA <sub>NED0</sub> value (X) shall be suitable for use as a conservative prediction of the RMS NED range errors for accuracy-related purposes in the pseudorange domain (e.g., measurement de- weighting RAIM, FOM computations). Integrity properties of the IAURA <sub>NED</sub> are specified with respect to the scaled (multiplied by either 4.42 or 5.73 as appropriate) upper bound values of the URA <sub>NED0</sub> index, URA <sub>NED1</sub> index, and URA <sub>NED2</sub> index (see 20.3.3.1.1). URA <sub>NED0</sub> accounts for zeroth order SIS-contributions to user range error which include, but are not limited to, the following: LSB representation/truncation error; the net effect of clock correction polynomial error and code phase error in the transmitted signal for single-frequency L1C/A or single- frequency L2C users who correct the code phase as described in Section 20.3.3.3.1.1.1; the net effect		The nominal URA <sub>NEDO</sub> value (X) shall be suitable for use as a conservative prediction of the RMS NED range errors for accuracy-related purposes in the pseudorange domain (e.g., measurement de-weighting RAIM, FOM computations). Integrity properties of the IAURA <sub>NED</sub> are specified with respect to the scaled (multiplied by either 4.42 or 5.73 as appropriate) upper bound values of the URA <sub>NEDO</sub> index, URA <sub>NED1</sub> index, and URA <sub>NED2</sub> index (see 20.3.3.1.1). URA <sub>NED0</sub> accounts for zeroth order SIS-contributions to user range error which include, but are not limited to, the following: LSB representation/truncation error; the net effect of clock correction polynomial error and code phase error in the transmitted signal for single-frequency L1C/A or single-frequency L2C users who correct the code phase as described in	relationship t the URA <sub>oc2</sub> :" to "The transmitted
	of clock parameter, code phase, and inter-signal correction error for dual-frequency L1/L2 and L1/L5 users who correct for group delay and ionospheric effects as described in Section 20.3.3.3.1.1.2; radial ephemeris error; anisotropic antenna errors; and signal deformation error. URA <sub>NED</sub> does not account for user range contributions due to the inaccuracy of the broadcast ionospheric data parameters used in the single-frequency ionospheric model or for other atmospheric effects. The transmitted URA <sub>NED1</sub> index is an integer value in the range 0 to 7. The URA <sub>NED1</sub> index has the following relationship to the URA <sub>-NED1</sub> value:		Section 20.3.3.3.1.1.1; the net effect of clock parameter, code phase, and inter-signal correction error for dual-frequency L1/L2 and L1/L5 users who correct for group delay and ionospheric effects as described in Section 20.3.3.3.1.1.2; radial ephemeris error; anisotropic antenna errors; and signal deformation error. URA <sub>NED</sub> does not account for user range contributions due to the inaccuracy of the broadcast ionospheric data parameters used in the single-frequency ionospheric model or for other atmospheric effects. The transmitted URA <sub>NED1</sub> index is an integer value in the range 0 to 7. The URA <sub>-NED1</sub> index has the following relationship to the URA <sub>-NED1</sub> value:	URA <sub>NED2</sub> index is an integer value in the range 0 to 7. URA <sub>NED2</sub> index has the following relationship t the URA <sub>NED2</sub> :"
	$URA_{NED1} = \overline{2^{N}}$ (meters/second) where		$URA_{NED1} = \frac{1}{2^{N}} $ (meters/second) where	

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	N = 14 + URA <sub>NED1</sub> Index.		N = 14 + URA <sub>NED1</sub> Index.	
	The transmitted $URA_{NED2}$ index is an integer value in the range 0 to 7. $URA_{NED2}$ index has the following relationship to the $URA_{oc2}$ :		The transmitted URA <sub>NED2</sub> index is an integer value in the range 0 to 7. URA <sub>NED2</sub> index has the following relationship to the URA <sub>NED2</sub> :	
	$URA_{NED2} = \frac{1}{2^{N}} \text{ (meters/second/second)}$ where		URA <sub>NED2</sub> = $\frac{1}{2^{N}}$ (meters/second/second)	
			where	
	$N = 28 + URA_{NED2} Index.$		N = 28 + URA <sub>NED2</sub> Index.	
20.3.3.4.4	The three, one-bit, health indication in bits 155, 156 and 157 of message type 37 and bits 29,30 and 31 of each packet of reduced almanac refers to the L1, L2, and L5 signals of the SV whose PRN number is specified in the message or in the packet. For each health indicator, a "O" signifies that all navigation data are okay and "1" signifies that some or all navigation data are bad. The predicted health data will be updated at the time of upload when a new reduced almanac has been built by the CS. The transmitted health data may not correspond to the actual health of the transmitting SV or other SVs in the constellation.		The three, one-bit, health indication in bits 155, 156 and 157 of message type 37 and bits 29,30 and 31 of each packet of reduced almanac refers to the L1, L2, and L5 signals of the SV whose PRN number is specified in the message or in the packet. For each health indicator, a "0" signifies that all signals on the associated frequency are okay and "1" signifies that some or all signals on the associated frequency are bad. The predicted health data will be updated at the time of upload when a new reduced almanac has been built by the CS. The transmitted health data may not correspond to the actual health of the transmitting SV or other SVs in the constellation.	The current language states that "For each health indicator, a "O" signifies that all navigation data are okay and "1" signifies that some or all navigation data are bad." This language is misleading

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				in that it implies that one bit designated with a "1" means that all navigation data (L1, L2, and L5) are bad, which may not be true.
				Recommende d text clarifies that a "1" signifies that some or all signals on the associated frequency are bad.
20.3.3.4.6.1			A 6-bit value of "000000" in the PRN <sub>a</sub> field shall indicate that no further Status Words are contained in the remainder of the data block. In this event, all subsequent bits in the data block field shall be filler bits, i.e., alternating ones and zeros beginning with one.	This language is being supplied so that users now know how to interpret dummy SVs for the L5 signal.

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20.3.3.4.6.2 .1	$\begin{array}{ c c c c c c } \hline & & & & & & & & & & & & & & & & & & $		31 BITS         1       7       15       22       29       30       31         PRNa $\delta_A$ $\Omega_0$ $\Phi_0$	Publication error during Word export. Figure is now correct in Word/PDF. The bit counts for the data fields in the Reduced Almanac Packet diagram have been
	Figure 20-16. Reduced Almanac Packet Content		Figure 20-16. Reduced Almanac Packet Content	restored.
20.3.3.5.1	The EOP fields in the message type 32 contain the EOP data needed to construct the ECEF-to-ECI coordinate transformation. The user computes the ECEF position of the SV antenna phase center using the equations shown in Table 20-II. The full coordinate transformation for translating to the corresponding ECI SV antenna phase center position may be accomplished in accordance with the computations detailed in Chapter 5 of IERS Technical Note 36: IERS Conventions (2010) and equations for UT1, xp and yp as documented in Table 20-VIII. Figure 5.1 on page 73 of that document depicts the computational flow starting from GCRS (Geocentric Celestial Reference System) to ITRS (International Terrestrial Reference System). Ongoing WGS 84 re-adjustment at NGA and incorporating the 2010 IERS Conventions, are expected to bring Earth based coordinate agreement to within 2 cm. In the context of the Conventions via either the Celestial Intermediate Origin (CIO) based approach" or the "Equinox based approach". The EOP parameters for $\Delta$ UT1 are to be applied within the "Rotation to terrestrial system" process, and the parameters for x <sub>p</sub> and y <sub>p</sub> are applied in the "Rotation for polar motion" process. Users are advised that the broadcast message type 32 EOP parameters already account for zonal, diurnal and semidiurnal effects (described in Chapter 8 of the IERS Conventions (2010)), so these effects should not be further applied by the user.		The EOP fields in the message type 32 contain the EOP data needed to construct the ECEF- to-ECI coordinate transformation. The user computes the ECEF position of the SV antenna phase center using the equations shown in Table 20-II. The full coordinate transformation for translating to the corresponding ECI SV antenna phase center position may be accomplished in accordance with the computations detailed in Chapter 5 of IERS Technical Note 36: IERS Conventions (2010) and equations for UT1, xp and yp as documented in Table 20-VIII. Figure 5.1 on page 73 of that document depicts the computational flow starting from GCRS (Geocentric Celestial Reference System) to ITRS (International Terrestrial Reference System). Ongoing WGS 84 re-adjustment at NGA and incorporating the 2010 IERS Conventions, are expected to bring Earth based coordinate agreement to within 2 cm. In the context of the Conventions, the user may as a matter of convenience choose to implement the transformation computations via either the "Celestial Intermediate Origin (CIO) based approach" or the "Equinox based approach". The EOP parameters for $\Delta$ UT1 are to be applied within the "Rotation to terrestrial system" process, and the parameters for x <sub>p</sub> and y <sub>p</sub> are applied in the "Rotation for polar motion" process. Users are advised that the broadcast message type 32 EOP parameters already account for zonal, diurnal and semidiurnal effects (described in Chapter 8 of the IERS Conventions (2010)), so	Added leading quotation so that the CIO based approach is within a pair of quotations: "Celestial Intermediate Origin (CIO) based approach".

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	The relevant computations utilize elementary rotation matrices $R_i(\alpha)$ , where $\alpha$ is a positive rotation about the i <sup>th</sup> -axis ordinate, as follows:		these effects should not be further applied by the user. The relevant computations utilize elementary rotation matrices $R_i(\alpha)$ , where $\alpha$ is a positive rotation about the i <sup>th</sup> -axis ordinate, as follows:	
	$ R_{1}(\alpha) = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos(\alpha) & \sin(\alpha) \\ 0 & -\sin(\alpha) & \cos(\alpha) \end{bmatrix} , R_{2}(\alpha) = \begin{bmatrix} \cos(\alpha) & 0 & -\sin(\alpha) \\ 0 & 1 & 0 \\ \sin(\alpha) & 0 & \cos(\alpha) \end{bmatrix} $ $ R_{3}(\alpha) = \begin{bmatrix} \cos(\alpha) & \sin(\alpha) & 0 \\ -\sin(\alpha) & \cos(\alpha) & 0 \\ 0 & 0 & 1 \end{bmatrix} $		$ \begin{split} R_1(\alpha) &= \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos(\alpha) & \sin(\alpha) \\ 0 & -\sin(\alpha) & \cos(\alpha) \end{bmatrix} \ , \ \ R_2(\alpha) &= \begin{bmatrix} \cos(\alpha) & 0 & -\sin(\alpha) \\ 0 & 1 & 0 \\ \sin(\alpha) & 0 & \cos(\alpha) \end{bmatrix} \\ R_3(\alpha) &= \begin{bmatrix} \cos(\alpha) & \sin(\alpha) & 0 \\ -\sin(\alpha) & \cos(\alpha) & 0 \\ 0 & 0 & 1 \end{bmatrix} \end{split} $	
20.3.3.7.2	Each DC data packet contains: corrections to SV clock polynomial coefficients provided in any one of the message types 30 to 37 of the corresponding SV; corrections to quasi-Keplerian elements referenced to $t_{OD}$ of the corresponding SV; User Differential Range Accuracy (UDRA) and UDRA indices that enable users to estimate the accuracy obtained after corrections are applied. Each DC packet is made up of two different segments. The first segment contains 34 bits for the CDC parameters and the second segment contains 92 bits of EDC parameters totaling 126 bits. The CDC and EDC parameters form an indivisible pair and users must utilize CDC and EDC as a pair. Users must utilize CDC and EDC data pairs of the same $t_{op-D}$ ( $t_{op-D}$ =DC data predict time of week) and of the same $t_{OD}$ .		Each DC data packet contains: corrections to SV clock polynomial coefficients provided in any one of the message types 30 to 37 of the corresponding SV; corrections to quasi- Keplerian elements referenced to $t_{oD}$ of the corresponding SV; User Differential Range Accuracy (UDRA) and $UDRA$ indices that enable users to estimate the accuracy obtained after corrections are applied. Each DC packet is made up of two different segments. The first segment contains 34 bits for the CDC parameters and the second segment contains 92 bits of EDC parameters totaling 126 bits. The CDC and EDC parameters form an indivisible pair and users must utilize CDC and EDC as a pair. Users must utilize CDC and EDC data pairs of the same $t_{op-D}$ ( $t_{op-D}$ =DC data predict time of week) and of the same $t_{OD}$ .	Incorporated DOT over proper UDRA term.
20.3.3.7.4	The user will construct a set of initial (uncorrected) elements by: $\begin{array}{rcl} A_{i} & = & A_{0} \\ e_{i} & = & e_{n} \\ i_{i} & = & i_{0-n} \\ \Omega_{i} & = & \Omega_{0-n} \\ \alpha_{i} & = & e_{n} \bullet \cos(\omega_{n}) \\ \beta_{i} & = & e_{n} \bullet \sin(\omega_{n}) \end{array}$		The user will construct a set of initial (uncorrected) elements by: $\begin{array}{rcl} A_{i} & = & A_{0} \\ e_{i} & = & e_{n} \\ i_{i} & = & i_{0-n} \\ \Omega_{i} & = & \Omega_{0-n} \\ \alpha_{i} & = & e_{n} \bullet \cos(\omega_{n}) \\ \beta_{i} & = & e_{n} \bullet \sin(\omega_{n}) \end{array}$	The current mean anomaly equation, $\Delta M_0$ , yields a velocity component and is

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	$\gamma_i = M_{0-n} + \omega_n$ where $A_0$ , $e_n$ , $i_{0-n}$ , $\Omega_{0-n}$ , $\omega_n$ and $M_{0-n}$ are obtained from the applicable SV's message types 10 and 11 data. The terms $\alpha_i$ , $\beta_i$ , and $\gamma_i$ form a subset of stabilized ephemeris elements which are subsequently corrected by $\Delta \alpha$ , $\Delta \beta$ and $\Delta \gamma$ —the values of which are supplied in the message types 34 or 14-as follows: $\alpha_c = \alpha_i + \Delta \alpha$		$\gamma_i = M_{0-n} + \omega_n$ where $A_0$ , $e_n$ , $i_{0-n}$ , $\Omega_{0-n}$ , $\omega_n$ and $M_{0-n}$ are obtained from the applicable SV's message types 10 and 11 data. The terms $\alpha_i$ , $\beta_i$ , and $\gamma_i$ form a subset of stabilized ephemeris elements which are subsequently corrected by $\Delta \alpha$ , $\Delta \beta$ and $\Delta \gamma$ —the values of which are supplied in the message types 34 or 14-as follows: $\alpha_c = \alpha_i + \Delta \alpha$	incorrect. The mean anomaly equation should yield 'radians.'
	$ \begin{array}{llllllllllllllllllllllllllllllllllll$		$\beta_{c} = \beta_{i} + \Delta\beta$ $\gamma_{c} = \gamma_{i} + \Delta\gamma$	
	The quasi-Keplerian elements are then corrected by $A_{c} = A_{i} + \Delta A$ $e_{c} = (\alpha_{c}^{2} + \beta_{c}^{2})^{1/2}$ $i_{c} = i_{i} + \Delta i$ $\Omega_{c} = \Omega_{i} + \Delta \Omega$		The quasi-Keplerian elements are then corrected by $A_{c} = A_{i} + \Delta A$ $e_{c} = (\alpha_{c}^{2} + \beta_{c}^{2})^{1/2}$ $i_{c} = i_{i} + \Delta i$ $\Omega_{c} = \Omega_{i} + \Delta \Omega$	
	$\begin{split} \omega_c &= \tan^{-1}\left(\beta_c/\alpha_c\right) \\ M_{0\_c} &= \gamma_c - \omega_c + \Delta M_0 \end{split}$ where $\Delta A$ , $\Delta i$ and $\Delta \Omega$ are provided in the EDC data packet of the message type 34 or 14 and $\Delta M_0$ is obtained from		$ \begin{split} \omega_c &= tan^{-1} \left(\beta_c / \alpha_c\right) \\ M_{0\_c} &= \gamma_c - \omega_c + \Delta M_0 \end{split}  \\                                 $	
	$\Delta M_0 = -3^* (\mu^{1/2}) / A_c^{2*} [(t_{oe}) - (t_{OD})].$ The corrected quasi-Keplerian elements above are applied to the user algorithm for determination of		$\Delta M_0 = \frac{-3}{2} \left(\frac{\mu}{A_0^3}\right)^{\frac{1}{2}} \left(\frac{\Delta A_0}{A_0}\right) [(t_{oe} + WN_{oe} * 604,800) - (t_{OD} + WN * 604,800)]$	
	antenna phase center position in Section 20.3.3.1.3, Table 20-II.		The corrected quasi-Keplerian elements above are applied to the user algorithm for determination of antenna phase center position in Section 20.3.3.1.3, Table 20-II.	

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20.3.3.7.5	The UDRA <sub>op-D</sub> and UDRA shall give the differential user range accuracy for the SV. It must be noted that the two parameters provide estimated accuracy after both clock and ephemeris DC are applied. The UDRA <sub>op-D</sub> and UDRA indices are signed, two's complement integers in the range of +15 to -16 and have the following relationship:		The UDRA <sub>op-D</sub> and $UDRA$ shall give the differential user range accuracy for the SV. It must be noted that the two parameters provide estimated accuracy after both clock and ephemeris DC are applied. The UDRA <sub>op-D</sub> and $UDRA$ indices are signed, two's complement integers in the range of +15 to -16 and have the following relationship:	Incorporated DOT over proper UDRA terms.

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20.3.3.7.5	Index Value	UDRA <sub>op-D</sub> (meters)	UDRA (10 <sup>-6</sup> m/sec)		Index Value		<u>UDRA<sub>o</sub></u>	<sub>p-D</sub> (meters)		UDRA (10	<sup>-6</sup> m/sec)	Incorporated DOT over
	15	$6144.00 < UDRA_{op-D} $ $6144.00$										proper UDRA
	14	$3072.00 < UDRA_{op-D} \le 6144.00$	3072.00									terms.
	13	$1536.00 < UDRA_{op-D} \le 3072.00$	1536.00		15	6144.00	<	UDRA <sub>op-D</sub>		6144.00		
	12	$768.00 < UDRA_{op-D} \le 1536.00$	768.00		1.4	2072.00			/	C144.00	2072.00	
	11	$384.00 < UDRA_{op-D} \leq 768.00$	384.00		14	3072.00	<	UDRA <sub>op-D</sub>	≤	6144.00	3072.00	
	10	$192.00 < UDRA_{op-D} \le 384.00$	192.00		13	1536.00	<	UDRA <sub>op-D</sub>	$\leq$	3072.00	1536.00	
	,	$96.00 < UDRA_{op-D} \le 192.00$	96.00		15	1330.00		ODIA op-D		5072.00	1550.00	
	8	$48.00 < UDRA_{op-D} \le 96.00$	48.00		12	768.00 <		<sub>p-D</sub> ≤	1536	.00 768	3.00	
		$24.00 < UDRA_{op-D} \le 48.00$	24.00				• • • • •	p-D —	2000			
	5	$13.65 < UDRA_{op-D} \le 24.00$ $9.65 < UDRA_{op-D} \le 13.65$	13.65		11	384.00 <		p-D ≤	768.0	00 384.00		
	3	$9.65 < UDRA_{op-D} \le 13.65$ $6.85 < UDRA_{op-D} \le 9.65$	9.65 6.85									
	3	$4.85 < UDRA_{op-D} \le 6.85$	4.85		10	192.00 <		p-D ≤	384.0	00 192.00		
	2	$3.40 < UDRA_{op-D} \le 4.85$	3.40									
	1	$2.40 < UDRA_{op-D} \le 3.40$	2.40		9	96.00 <		p-D ≤	192.0	00 96.00		
	0	$1.70 < UDRA_{op-D} \le 2.40$	1.70									
	-1	$1.20 < UDRA_{op-D} \le 1.70$	1.20		8	48.00 <	UDRA <sub>o</sub>	p-D ≤	96.00	) 48.00		
	-2	$0.85 < UDRA_{op-D} \le 1.20$	0.85		7	24.00 <			10 00	24.00		
	-3	$0.60 < UDRA_{op-D} \le 0.85$	0.60		/	24.00 <		<sub>p-D</sub> ≤	40.00	) 24.00		
	-4	$0.43 < UDRA_{op-D} \leq 0.60$	0.43		6	13.65 <		<sub>p-D</sub> ≤	24.00	) 13.65		
	-5	$0.30  <  \text{UDRA}_{\text{op-D}} \leq \qquad 0.43$	0.30		Ū	13.05	001010	p-D —	24.00	, 13.05		
l	-6	$0.21 < UDRA_{op-D} \leq 0.30$	0.21		5	9.65 <		p-D ≤	13.65	5 9.65		
l	-7	$0.15 < UDRA_{op-D} \le 0.21$	0.15				0	<i>p b</i>				
	-8	$0.11 < UDRA_{op-D} \le 0.15$	0.11		4	6.85 <		p-D ≤	9.65	6.85		
	-9	$0.08 < \text{UDRA}_{\text{op-D}} \leq 0.11$	0.08									
	-10	$0.06 < UDRA_{op-D} \le 0.08$	0.06		3	4.85 <		p-D ≤	6.85	4.85		
	-11	$0.04 < \text{UDRA}_{\text{op-D}} \leq 0.06$	0.04									
	-12	$0.03 < \text{UDRA}_{\text{op-D}} \leq 0.04$	0.03		2	3.40 <		p-D ≤	4.85	3.40		
1	-13	$0.02 < UDRA_{op-D} \leq 0.03$	0.02			2.40				2.40		
	-14	$0.01 < UDRA_{op-D} \leq 0.02$	0.01		1	2.40 <		p-D ≤	3.40	2.40		
	-15	$UDRA_{op-D} \leq 0.01$	0.005			170 -		/	2 40	1 70		
	-16	No accuracy prediction available—use	at own risk		0	1.70 <	UDRA <sub>o</sub>	p-D ≤	2.40	1.70		
					-1	1.20 <		p-D ≤	1.70	1.20		
					-2	0.85 <		p-D ≤	1.20	0.85		

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			-3	0.60	<	UDRA <sub>op-D</sub>	≤	0.85	0.60	
			-4	0.43	<	UDRA <sub>op-D</sub>	$\leq$	0.60	0.43	
			-5	0.30	<	UDRA <sub>op-D</sub>	$\leq$	0.43	0.30	
			-6	0.21	<	UDRA <sub>op-D</sub>	$\leq$	0.30	0.21	
			-7	0.15	<	UDRA <sub>op-D</sub>	≤	0.21	0.15	
			-8	0.11	<	UDRA <sub>op-D</sub>	$\leq$	0.15	0.11	
			-9	0.08	<	UDRA <sub>op-D</sub>	≤	0.11	0.08	
			-10	0.06	<	UDRA <sub>op-D</sub>	≤	0.08	0.06	
			-11	0.04	<	UDRA <sub>op-D</sub>	≤	0.06	0.04	
			-12	0.03	<	UDRA <sub>op-D</sub>	≤	0.04	0.03	
			-13	0.02	<	UDRA <sub>op-D</sub>	≤	0.03	0.02	
			-14	0.01	<	UDRA <sub>op-D</sub>	≤	0.02	0.01	
			-15			UDRA <sub>op-D</sub>	≤	0.01	0.005	
			-16	No a	ccuracy	prediction availa	ıble-use	at own ri	sk	
20.3.3.7.5	For any time, $t_k$ , other than $t_{op-D}$ , UDRA is found by,		For any time, $t_k$ , other than $t_{op-D}$ , UDRA is found by,						Incorporated DOT over proper UDRA	
	$UDRA = UDRA_{op-D} + UDRA (t_k - t_{op-D})$			A = UDR	م <sub>op-D</sub> + ۷	· UDRA (t <sub>k</sub> - t <sub>op-D</sub> )				term.

End of WAS/IS for IS-GPS-705B, IRN-001 Changes